

# Ch. 6. Momentum and Forces in Fluid Flow

## 6.1. Development of the Momentum Principle

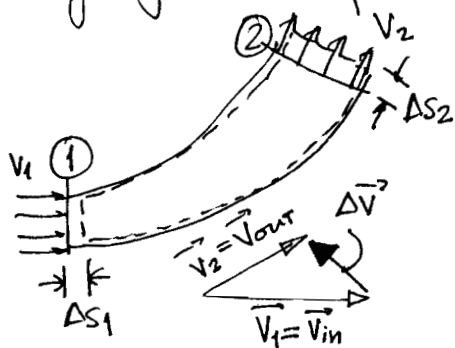
Start with a modified form of Newton's 2nd Law:  $\Sigma \vec{F} = \frac{d}{dt}(\underbrace{m\vec{v}}_{\text{on a system}})$

NOTE: If  $m = \text{constant}$ ,  $\frac{d}{dt}(m\vec{v}) = m \frac{d\vec{v}}{dt} = m\vec{a}$ .

Also,  $\Sigma \vec{F} dt = d(m\vec{v})$  ← impulse-momentum principle

$\uparrow$                        $\uparrow$   
 impulse =  $\Delta(\text{momentum})$

Using Reynolds Transport Theorem (section 4.6)



$$\frac{dX_s}{dt} = \frac{dX_{cv}}{dt} + \frac{dX_{cv}^{out}}{dt} - \frac{dX_{cv}^{in}}{dt} \quad \leftarrow \text{GENERAL UNSTEADY CASE}$$

with  $X = m\vec{v}$   
 for steady flow,  $\frac{dX_{cv}}{dt} = 0$

$$\frac{dX_s}{dt} = \frac{dX_{cv}^{out}}{dt} - \frac{dX_{cv}^{in}}{dt}$$

for momentum principle:

$$\Sigma \vec{F} = \frac{d}{dt}(m\vec{v})_{cv}^{out} - \frac{d}{dt}(m\vec{v})_{cv}^{in}$$

For steady flow the net force on the fluid mass is equal to the net rate of outflow of momentum across the control surface.

- SELECT a C.V. such that  $V \perp C.S.$  where it cuts the flow
- $V = \text{constant}$  where C.S. cuts flow  $\Rightarrow \frac{d}{dt}(m\vec{v})_1 = \frac{dm_1}{dt}\vec{v}_1 = \dot{m}_1\vec{v}_1 = \rho_1 Q_1 \vec{v}_1$

Thus, for steady flow  $\Sigma \vec{F} = \dot{m}_2 \vec{v}_2 - \dot{m}_1 \vec{v}_1 = \rho_2 Q_2 \vec{v}_2 - \rho_1 Q_1 \vec{v}_1$

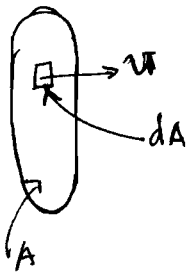
Since the flow is steady  $\dot{m}_1 = \dot{m}_2 = \dot{m} = \rho_1 Q_1 = \rho_2 Q_2 = \rho Q \Rightarrow \rho_2 Q_2 \vec{v}_2 - \rho_1 Q_1 \vec{v}_1 = \rho Q (\vec{v}_2 - \vec{v}_1) = \rho Q \Delta \vec{v}$ , with  $\Delta \vec{v} = \vec{v}_2 - \vec{v}_1 = \vec{v}_{out} - \vec{v}_{in}$

$\Rightarrow$  MOMENTUM (or impulse-momentum) principle

$$\Sigma \vec{F} = \dot{m} \Delta \vec{v} = \rho Q \Delta \vec{v} = \rho Q (\vec{v}_2 - \vec{v}_1)$$



## 6.3. Momentum Correction Factor



$$\left\{ \begin{array}{l} \text{momentum flux} \\ \text{across } dA \end{array} \right\} = dm \cdot u = (\rho u dA) u = \rho u^2 dA$$

Volume  
 $dV = v dt dA$

$$\left\{ \begin{array}{l} \text{ACTUAL} \\ \text{momentum} \\ \text{flux across} \\ A \end{array} \right\} = \int_A \rho u^2 dA$$

mass  
 $dm = \rho dV = \rho v dt dA$

$$\left\{ \begin{array}{l} \text{momentum} \\ \text{flux with} \\ \text{average velocity} \end{array} \right\} = \int_A \rho v^2 dA = \rho v^2 A$$

$$\left\{ \begin{array}{l} \text{actual momentum} \\ \text{flux across } A \end{array} \right\} = \beta \cdot \left\{ \begin{array}{l} \text{momentum flux} \\ \text{with average velocity} \end{array} \right\}$$



$$\int_A \rho u^2 dA = \beta \rho v^2 A$$

with constant  $\rho$ :

$$\beta = \frac{1}{AV^2} \int_A u^2 dA$$

For laminar flow in circular pipe,  $\beta = \frac{4}{3} = 1.33$

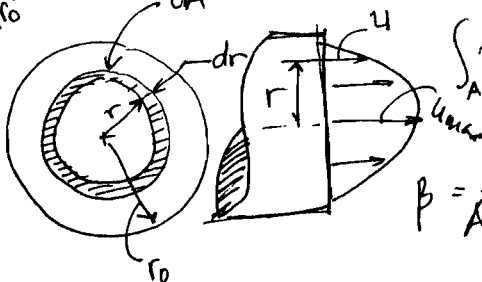
For turbulent flow in circular pipe,  $\beta = 1.005$  to  $1.05$ , i.e.,  $\beta \approx 1.0$

E.g. for laminar flow ~~(turbulent)~~  $Q = \int_A u dA = \int_0^{r_0} u_{max} \left(1 - \left(\frac{r}{r_0}\right)^2\right) \cdot 2\pi r dr$

$$u = u_{max} \left(1 - \left(\frac{r}{r_0}\right)^2\right)$$

$$Q = \frac{\pi}{2} u_{max} r_0^2, \quad V = \frac{Q}{A} = \frac{\frac{\pi}{2} u_{max} r_0^2}{2\pi r_0^2} = \frac{u_{max}}{2}$$

$$A = \pi r_0^2, \quad dA = 2\pi r dr$$



$$\int_A u^2 dA = \int_0^{r_0} u_{max}^2 \left(1 - \left(\frac{r}{r_0}\right)^2\right)^2 \cdot 2\pi r dr = \frac{u_{max}^2 \pi}{3}$$

$$\beta = \frac{1}{AV^2} \int_A u^2 dA = \frac{1}{\pi r_0^2 \frac{u_{max}^2}{4}} \cdot \frac{u_{max}^2 \pi r_0^2}{3} = \frac{4}{3} \leftarrow \begin{array}{l} \text{momentum} \\ \text{correction} \\ \text{factor} \end{array}$$

$$\text{also, } \int_A u^3 dA = \int_0^{r_0} u_{max}^3 \left[1 - \left(\frac{r}{r_0}\right)^2\right]^3 \cdot 2\pi r dr = \frac{\pi u_{max}^3 r_0^2}{4}, \quad \alpha = \frac{1}{AV^3} \int_A u^3 dA = 2.0$$

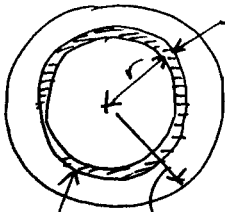
$\leftarrow$  kinetic energy correction factor

One turbulent flow possibility in pipes (see eq. 8.49)

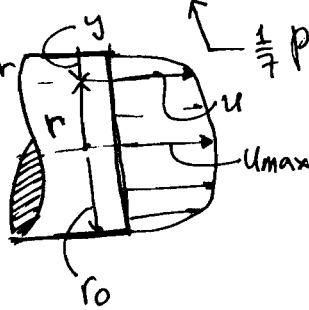
Smooth pipe,

$$\frac{u}{u_{\max}} = \left(\frac{y}{r_0}\right)^{1/7}, \quad y = \text{measured from wall}$$

$$A = \pi r_0^2$$



$$dA = 2\pi r dr$$



$\frac{1}{7}$  power law

NOTE:  $r+y=r_0$ , thus

$$u = u_{\max} \left(\frac{r_0 - r}{r_0}\right)^{1/7} \quad \text{or}$$

$$u = u_{\max} \left(1 - \frac{r}{r_0}\right)^{1/7}$$

$$Q = \int_A u dA = \int_0^{r_0} u_{\max} \left(1 - \frac{r}{r_0}\right)^{1/7} 2\pi r dr = \frac{49}{60} \pi u_{\max} r_0^2, \quad V = \frac{Q}{A} = \frac{\frac{49}{60} \pi u_{\max} r_0^2}{\pi r_0^2}$$

$$\int u^2 dA = \int_0^{r_0} u_{\max}^2 \left(1 - \frac{r}{r_0}\right)^{2/7} 2\pi r dr = \frac{49}{72} \pi u_{\max}^2 r_0^2$$

$$V = \frac{49}{60} u_{\max}$$

$$\beta = \frac{1}{AV^2} \int u^2 dA = \frac{1}{\pi r_0^2 \left(\frac{49}{60} u_{\max}\right)^2} \left(\frac{49}{72} \pi u_{\max}^2 r_0^2\right) = \frac{50}{49} = 1.020 \leftarrow \begin{array}{l} \text{momentum correction} \\ \text{factor} \end{array}$$

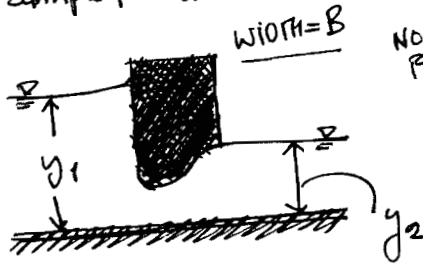
$$\int u^3 dA = \int_0^{r_0} u_{\max}^3 \left(1 - \frac{r}{r_0}\right)^{3/7} 2\pi r dr = \frac{49}{85} \pi u_{\max}^3 r_0^2$$

$$\alpha = \frac{1}{AV^3} \int u^3 dA = \frac{1}{\pi r_0^2 \left(\frac{49}{60} u_{\max}\right)^3} \cdot \frac{49}{85} \pi u_{\max}^3 r_0^2 = \frac{43200}{40817} = 1.058 \leftarrow$$

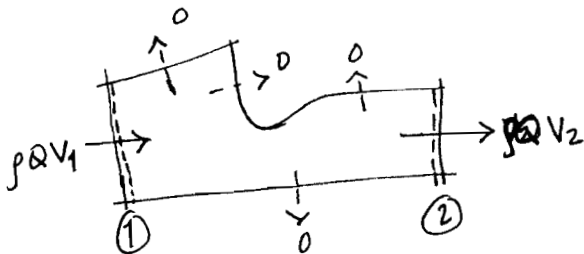
Kinetic energy correction factor

# 6.4. Applications of the Momentum Principle - Open-channel flow cases

See Sample Problem 6.1



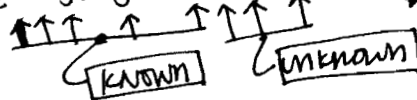
Momentum Fluxes



Momentum Principle components

$$\Sigma F_x = \Delta(pQV)_x \Rightarrow \frac{1}{2} \gamma B (y_1^2 - y_2^2) - F_x = \rho Q (v_2 - v_1)$$

$$F_x = \frac{1}{2} \gamma B (y_1^2 - y_2^2) - \rho Q (v_2 - v_1)$$



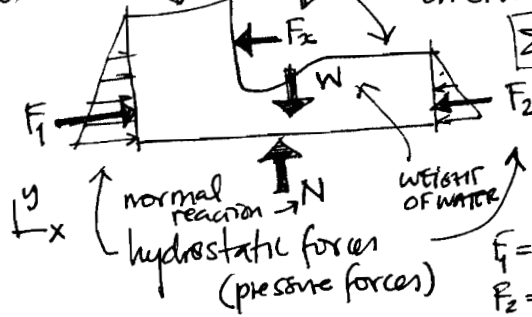
Use continuity:  $Q = \beta y_1 v_1 = \beta y_2 v_2 \Rightarrow y_1 v_1 = y_2 v_2$  } solve for  $v_1, v_2$   
 and energy:  $y_1 + \frac{v_1^2}{2g} = y_2 + \frac{v_2^2}{2g}$  } calculate  $Q$ , then

FORCES ON THE C.V.

$$\Sigma F_z = F_1 - F_2 - F_3$$

NO FORCES ON FREE SURFACE

REACTION FROM WALL ON C.V.



$$\Sigma F_y = N - W$$

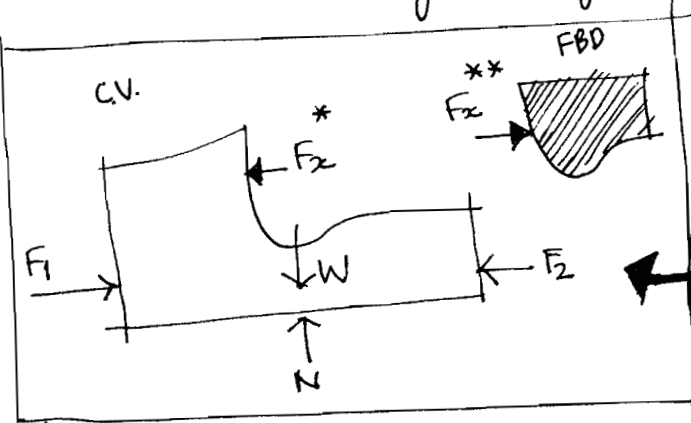
$$\Delta(pQV)_z = \rho Q v_2 - \rho Q v_1$$

$$\Delta(pQV)_y = 0 \text{ [NO FLOW IN VERTICAL DIRECTION]}$$

$$\Sigma F_y = \Delta(pQV)_y$$

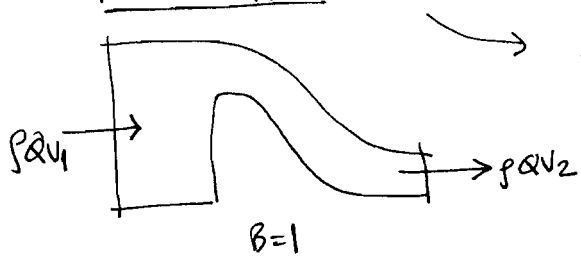
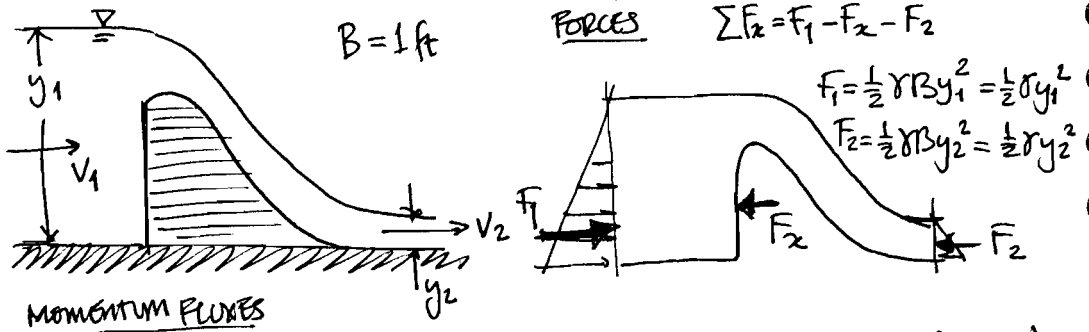
$$N - N = 0 \Rightarrow N = W$$

\*  $\leftarrow F_x$ : force applied by obstacle on C.V.  
 \*\*  $F_x \rightarrow$ : force applied by flow on obstacle



EXERCISES.

6.4.4. Flow occurs over a spillway of constant section as shown in Fig. X6.4.4. Assuming ideal flow, determine the resultant horizontal force on the spillway per foot of spillway width (perpendicular to the spillway section), given that  $y_1 = 4.2$  ft and  $y_2 = 0.7$  ft.



$$\Delta(pQV)_x = \rho Q V_2 - \rho Q V_1 = \rho Q (V_2 - V_1)$$

impulse-momentum:  $\Sigma F_x = \Delta(pQV)_x$

$$\frac{1}{2} \rho y_1^2 V_1 - F_x - \frac{1}{2} \rho y_2^2 V_2 = \rho Q (V_2 - V_1)$$

$$F_x = \frac{\rho}{2} (y_1^2 V_1 - y_2^2 V_2) - \rho Q (V_2 - V_1)$$

continuity:  $Q = B y_1 V_1 = B y_2 V_2$

$$y_1 V_1 = y_2 V_2 \Rightarrow \frac{V_2}{V_1} = \frac{y_1}{y_2}$$

energy:  $y_1 + \frac{V_1^2}{2g} = y_2 + \frac{V_2^2}{2g} \Rightarrow \frac{V_1^2}{2g} - \frac{V_2^2}{2g} = y_2 - y_1 \Rightarrow \frac{V_1^2}{2g} [1 - (\frac{y_2}{y_1})^2] = y_2 - y_1$

$$\Rightarrow \frac{V_1^2}{2g} [1 - (\frac{y_2}{y_1})^2] = y_2 - y_1 \Rightarrow V_1 = \sqrt{\frac{2g(y_2 - y_1)}{1 - (y_2/y_1)^2}}$$

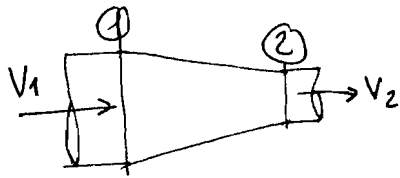
$$V_1 = \sqrt{\frac{2 \times 32.2 \text{ ft/s}^2 \times (0.7 - 4.2) \text{ ft}}{1 - (4.2/0.7)^2}} = 2.54 \text{ ft/s}, \quad V_2 = \frac{y_1}{y_2} V_1 = \frac{4.2}{0.7} \times 2.54 = 15.22 \frac{\text{ft}}{\text{s}}$$

$$Q = B y_1 V_1 = (1 \text{ ft})(4.2 \text{ ft})(2.54 \frac{\text{ft}}{\text{s}}) = 10.668 \text{ ft}^3/\text{s}$$

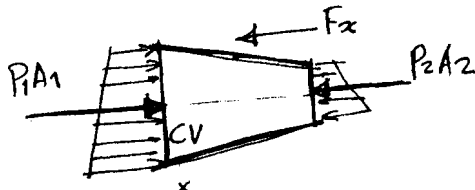
$$F_x = \frac{62.4 \text{ lb/ft}^3}{2} \times (4.2^2 - 0.7^2) \text{ ft}^2 \times 1 \text{ ft} - \frac{62.4 \text{ lb/ft}^3}{32.2 \text{ ft/s}^2} \times (15.22 - 2.54) \frac{\text{ft}}{\text{s}} \times 10.668 \frac{\text{ft}^3}{\text{s}}$$

$F_x = 272.94 \text{ lb}$

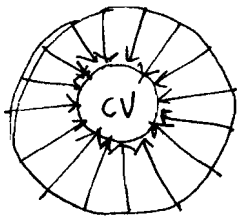
# 6.5. Force on Pressure Conduit e.g. reducer



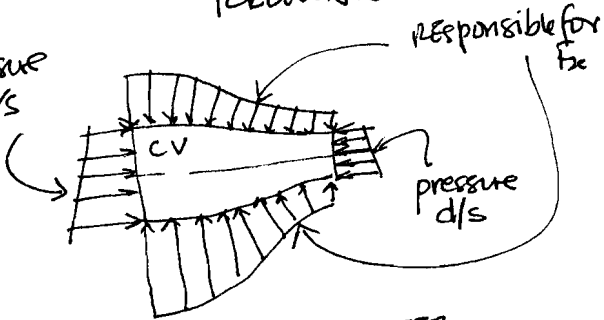
↑  
FLOW THROUGH REDUCER



↑  
FORCES ON CONTROL VOLUME



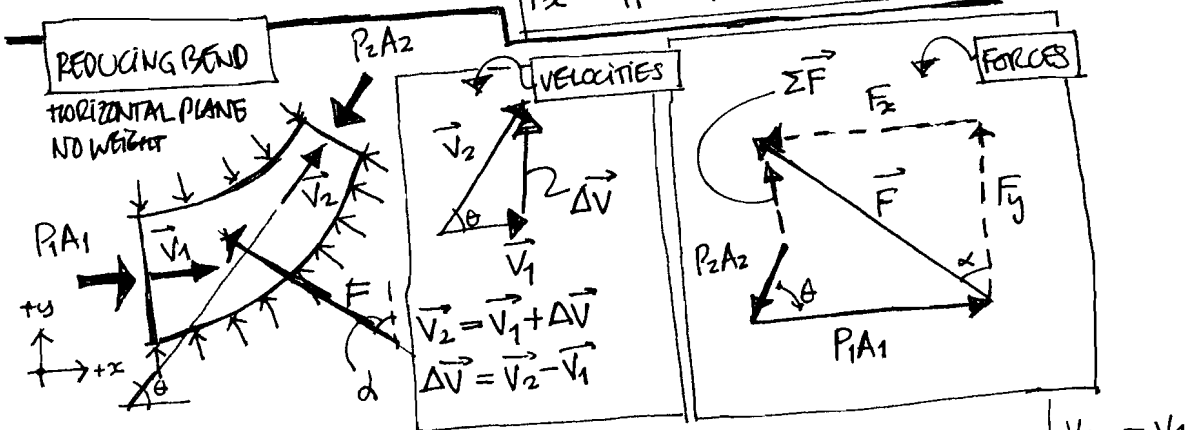
pressure u/s



↗ PRESSURE DISTRIBUTION ON THE FWD IN A REDUCER

$$\Sigma F_x = \Delta(\rho Q V)_x \Rightarrow P_1 A_1 - F_x - P_2 A_2 = \rho Q (V_2 - V_1)$$

$$F_x = P_1 A_1 - P_2 A_2 - \rho Q (V_2 - V_1)$$



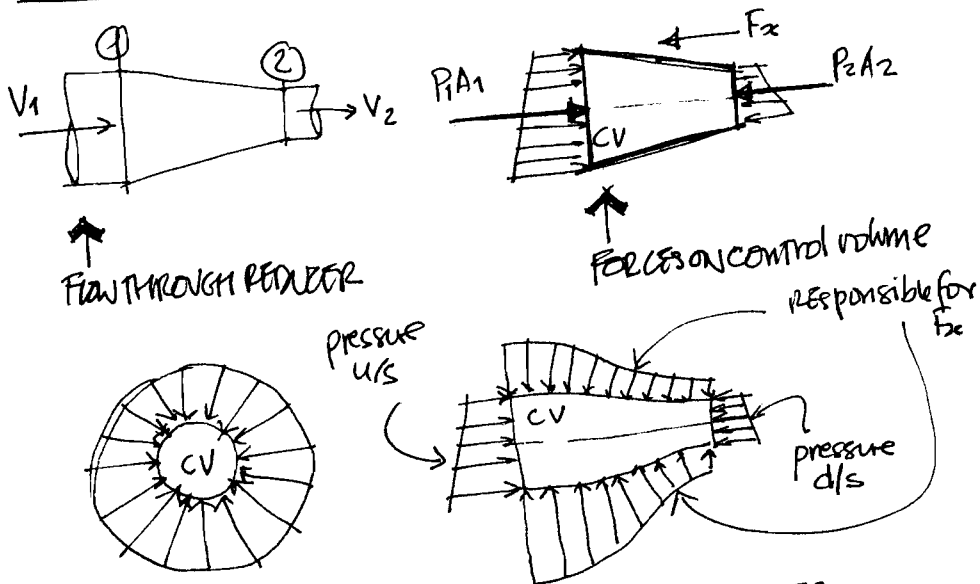
x-component:  $\Sigma F_x = \Delta(\rho Q V)_x \Rightarrow P_1 A_1 - F_x - P_2 A_2 \cos \theta = \rho Q (V_2 \cos \theta - V_1)$   $V_{1x} = V_1$   
 $V_{2x} = V_2 \cos \theta$

$$F_x = P_1 A_1 - P_2 A_2 \cos \theta - \rho Q (V_2 \cos \theta - V_1)$$

y-component:  $\Sigma F_y = \Delta(\rho Q V)_y \Rightarrow -P_2 A_2 \sin \theta + F_y = \rho Q (V_2 \sin \theta - V_{1y})$   $V_{1y} = 0$   
 $V_{2y} = V_2 \sin \theta$

$$F_y = P_2 A_2 \sin \theta + \rho Q V_2 \sin \theta$$

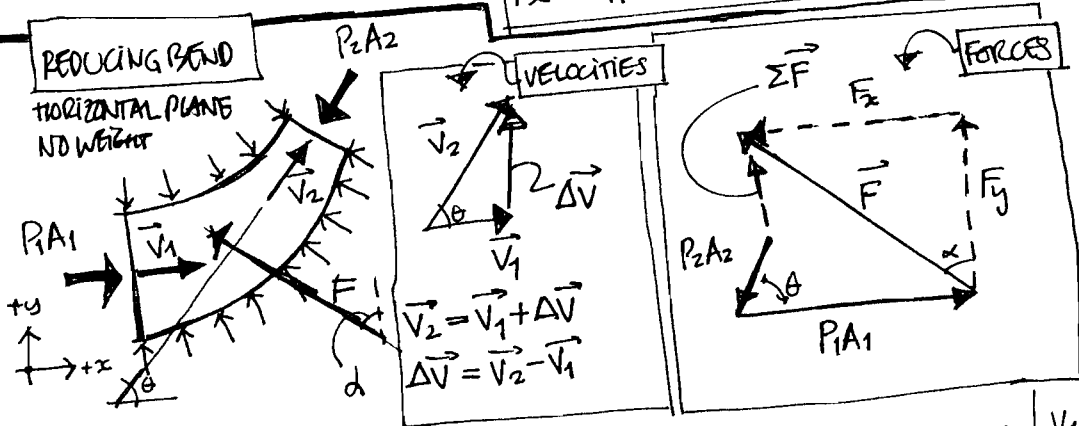
# 6.5. Forces on Pressure Conduit e.g. reducer



▲ PRESSURE DISTRIBUTION ON THE FWD IN A REDUCER

$$\Sigma F_x = \Delta(\rho Q V)_x \Rightarrow P_1 A_1 - F_x - P_2 A_2 = \rho Q (V_2 - V_1)$$

$$F_x = P_1 A_1 - P_2 A_2 - \rho Q (V_2 - V_1)$$



x-component:  $\Sigma F_x = \Delta(\rho Q V)_x \Rightarrow P_1 A_1 - F_x - P_2 A_2 \cos \theta = \rho Q (V_2 \cos \theta - V_1)$   $V_{1x} = V_1$   
 $V_{2x} = V_2 \cos \theta$

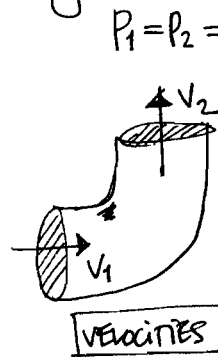
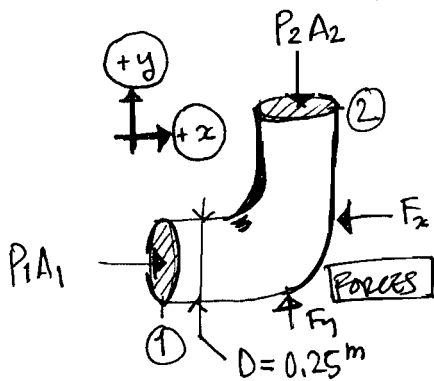
$$F_x = P_1 A_1 - P_2 A_2 \cos \theta - \rho Q (V_2 \cos \theta - V_1)$$

y-component:  $\Sigma F_y = \Delta(\rho Q V)_y \Rightarrow -P_2 A_2 \sin \theta + F_y = \rho Q (V_2 \sin \theta - V_{1y})$   $V_{1y} = 0$   
 $V_{2y} = V_2 \sin \theta$

$$F_y = P_2 A_2 \sin \theta + \rho Q V_2 \sin \theta$$

## EXERCISES

6.5.5. Water under a gage pressure of 350 kPa flows with a velocity of 5 m/s through a right-angled bend that has a uniform diameter of 250 mm. The bend lies in a horizontal plane, and water enters from the west and leaves toward the north. Assuming no drop in pressure, what is the magnitude and direction of the resultant force acting on the bend?



$$P_1 = P_2 = 350 \text{ kPa}$$

$$V_1 = V_2 = 5 \text{ m/s}$$

$$A_1 = A_2 = \frac{\pi D^2}{4} = \frac{\pi \cdot 0.25^2}{4} \\ = 4.91 \times 10^{-2} \text{ m}^2$$

$$\rightarrow \oplus \sum F_x = \Delta(pQV)_x \Rightarrow +P_1 A_1 = pQ(V_{2x} - V_{1x}) = pQ(0 - V_1)$$

$$F_x = P_1 A_1 + pQV_1 = (350 \times 10^3 \frac{\text{N}}{\text{m}^2})(4.91 \times 10^{-2} \text{ m}^2) +$$

$$(1000 \frac{\text{kg}}{\text{m}^3})(0.2454 \frac{\text{m}^3}{\text{s}})(5 \text{ m/s})$$

$$Q = V \frac{\pi D^2}{4} = (5 \frac{\text{m}}{\text{s}}) \cdot \frac{\pi \cdot (0.25 \text{ m})^2}{4}$$

$$Q = 0.2454 \text{ m}^3/\text{s}$$

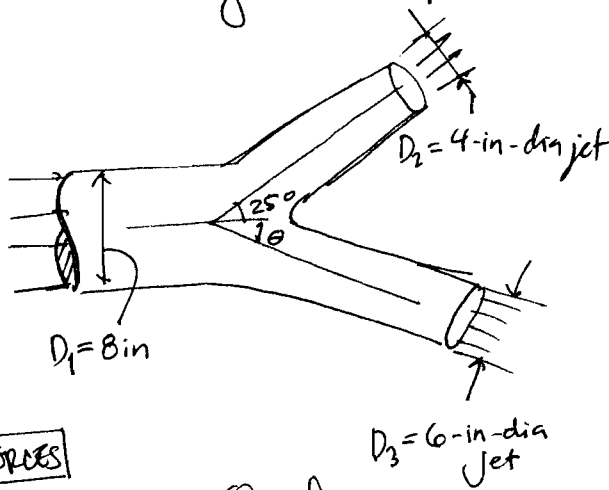
$$F_x = 18412 \text{ N} = 18.41 \text{ kN}$$

$$\uparrow \sum F_y = \Delta(pQV)_y \Rightarrow F_y - P_2 A_2 = pQ(V_{2y} - V_{1y}) = pQ(V_2 - 0)$$

$$F_y = P_2 A_2 + pQV_2 = (350 \times 10^3 \frac{\text{N}}{\text{m}^2})(4.91 \times 10^{-2} \text{ m}^2) + (1000 \frac{\text{kg}}{\text{m}^3})(0.2454 \frac{\text{m}^3}{\text{s}})(5 \text{ m/s})$$

$$F_y = 18412 \text{ N} = 18.41 \text{ kN}$$

6.5.3. In Fig. X6.5.3, both nozzles discharge water horizontally into the atmosphere at 30 fps. Find  $\theta$  so that the resultant force on the unit is along the axis of the 8-in-diameter pipe.



$$A_1 = \frac{\pi(8/12)^2}{4} = 0.349 \text{ ft}^2$$

$$A_2 = \frac{\pi(4/12)^2}{4} = 0.087 \text{ ft}^2$$

$$A_3 = \frac{\pi(6/12)^2}{4} = 0.196 \text{ ft}^2$$

continuity

$$Q_1 = Q_2 + Q_3$$

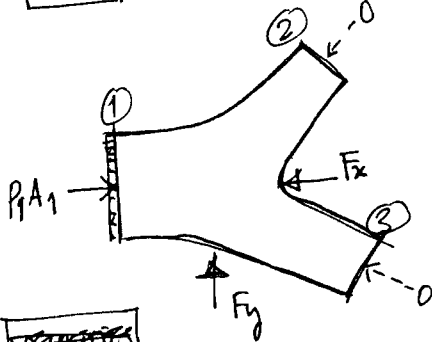
$$V_1 \cdot \frac{\pi D_1^2}{4} = V_2 \cdot \frac{\pi D_2^2}{4} + V_3 \cdot \frac{\pi D_3^2}{4}$$

$$V_1 D_1^2 = V_2 D_2^2 + V_3 D_3^2$$

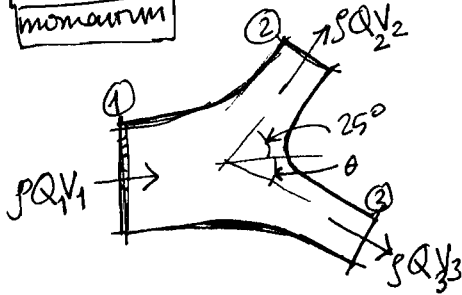
$$64V_1 = 16V_2 + 36V_3$$

$$\boxed{16V_1 = 4V_2 + 9V_3}$$

FORCES



momentum



momentum in x:

$$P_1 A_1 - F_x = \rho(Q_2 V_2 \cos 25^\circ + Q_3 V_3 \cos \theta - Q_1 V_1)$$

$$QV = (V \cdot A) \cdot V = V^2 A$$

momentum in y:

$$F_y = \rho(Q_2 V_2 \sin 25^\circ - Q_3 V_3 \sin \theta) = 0$$

$$\boxed{F_y = \rho A_2 V_2^2 \sin 25^\circ - \rho A_3 V_3^2 \sin \theta = 0}$$

energy  $v_1 = v_2 = v_3$

$$\frac{P_1}{\rho} + \frac{V_1^2}{2g} = \frac{V_2^2}{2g}$$

$$\frac{P_1}{\rho} + \frac{V_1^2}{2g} = \frac{V_3^2}{2g}$$

$$V_2 = V_3$$

$$\sin \theta = \frac{A_2 (V_2)^2 \sin 25^\circ}{A_3 (V_3)^2}$$

$$\sin \theta = \frac{0.087}{0.196} \sin 25^\circ = 0.1076$$

$$\boxed{\theta = 10.81^\circ}$$

