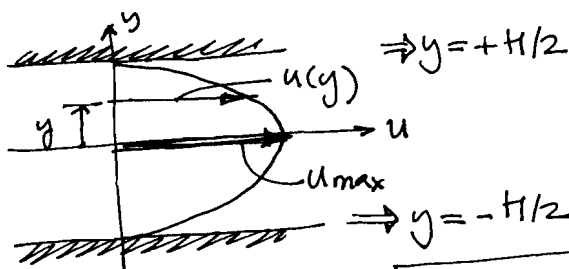
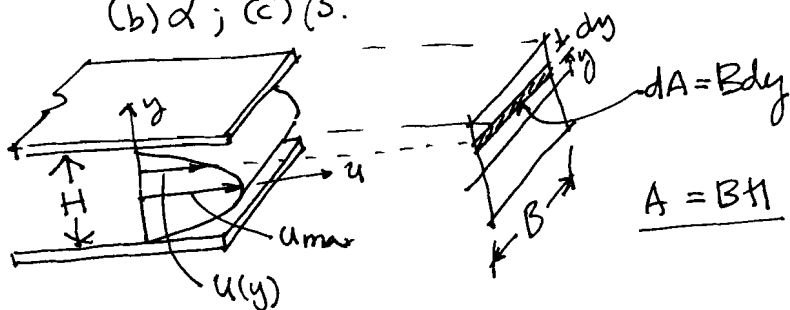


- 6.1. Two-dimensional laminar flow between two stationary parallel plates (parabolic velocity distribution). Find (a) $V/u_{max} = ?$ (b) α ; (c) β .



parabolic distribution:
$$u(y) = u_{max} \left(1 - \left(\frac{y}{H/2} \right)^2 \right)$$

check: when $y = 0 \rightarrow u(y) = u_{max}$
when $y = \pm H/2 \rightarrow u(y) = 0$

$$Q = \int_A u dA = \int_{-H/2}^{+H/2} u_{max} \left[1 - \left(\frac{2y}{H} \right)^2 \right] \cdot B \cdot dy = \frac{2HB u_{max}}{3}$$

$$V = \frac{Q}{A} = \frac{\frac{2}{3} HB u_{max}}{BH} = \frac{2}{3} u_{max} \Rightarrow (a) \frac{V}{u_{max}} = \frac{2}{3} = 0.666$$

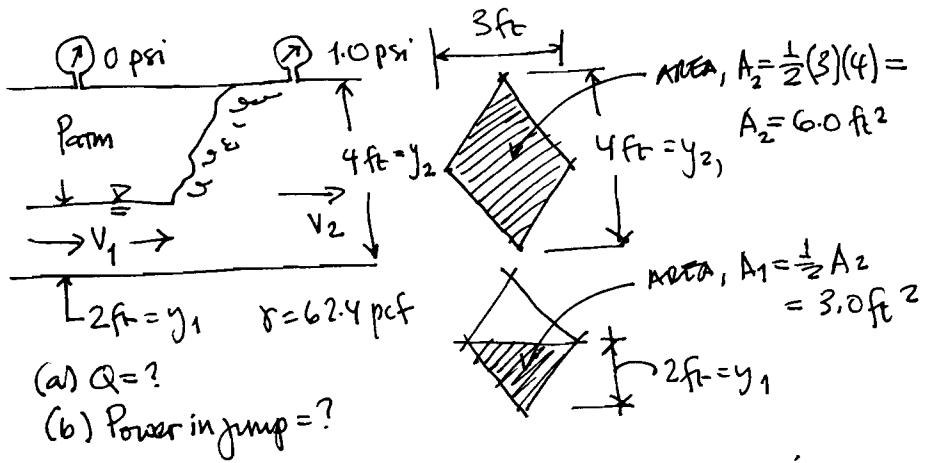
$$\int_A u^2 dA = \int_{-H/2}^{H/2} u_{max}^2 \left[1 - \left(\frac{2y}{H} \right)^2 \right]^2 \cdot B dy = \frac{8}{15} B \cdot H \cdot u_{max}^2$$

$$\int_A u^3 dA = \int_{-H/2}^{H/2} u_{max}^3 \left[1 - \left(\frac{2y}{H} \right)^2 \right]^3 \cdot B dy = \frac{16}{35} B \cdot H \cdot u_{max}^3$$

$$(b) \alpha = \frac{1}{AV^3} \int_A u^3 dA = \frac{1}{B \cdot H \cdot \left(\frac{2}{3} u_{max} \right)^3} \cdot \frac{16}{35} B \cdot H \cdot u_{max}^3 = \frac{54}{35} = 1.54$$

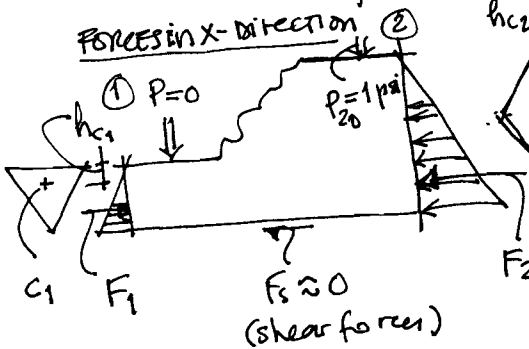
$$(c) \beta = \frac{1}{AV^2} \int_A u^2 dA = \frac{1}{B \cdot H \cdot \left(\frac{2}{3} u_{max} \right)^2} \cdot \frac{8}{15} B \cdot H \cdot u_{max}^2 = \frac{6}{5} = 1.20$$

6.4. hydraulic jump in close conduit (diamond-shape)



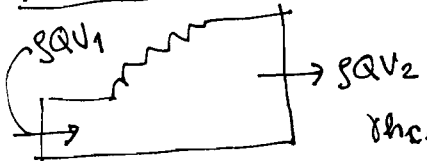
- Turbulence in jump \Rightarrow can not use energy equation
- use momentum equation

FORCES IN X-DIRECTION



$hc_1 = \frac{1}{3}(y_1)$
 $hc_2 = \frac{1}{2}y_2$
 $F_1, F_2 = \text{hydrostatic forces}$
 $F_1 = (P_1)_c A_1 = \gamma hc_1 A_1$
 $F_2 = (P_2)_c A_2 = (P_{20} + \gamma hc_2) A_2$
 NOTE: SECTION (2) IS PRESSURIZED IN THE SURFACE

MOMENTUM FLUX IN X-DIRECTION



MOMENTUM PRINCIPLE IN X-DIRECTION

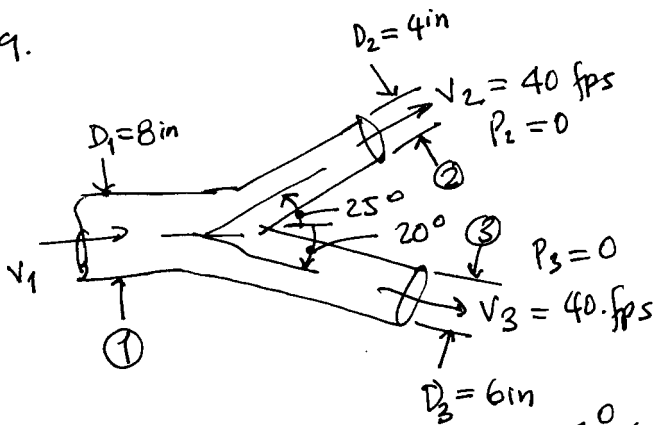
$\sum F_x = \Delta(\rho Q V)_x$
 $F_1 - F_2 - F_3 = \rho Q (V_2 - V_1)$

use $V_2 = \frac{Q}{A_2}$, $V_1 = \frac{Q}{A_1} \Rightarrow \gamma hc_1 A_1 - P_{20} A_2 - \gamma hc_2 A_2 = \rho Q^2 \left(\frac{1}{A_2} - \frac{1}{A_1} \right)$

$Q^2 = \frac{\gamma (hc_1 A_1 - hc_2 A_2) - P_{20} A_2}{\rho \left(\frac{1}{A_2} - \frac{1}{A_1} \right)} = \frac{62.4 \left(\frac{2}{3} \times 3 - \frac{4}{2} \times 6 \right) - 1 \times 144 \times 6}{32.2 \left(\frac{1}{6} - \frac{1}{3} \right)}$

$Q^2 = 4607.07 \text{ cfs}^2 \Rightarrow \boxed{Q = 67.87 \text{ cfs}}$

6.9.



$\rho = 62.4 \frac{\text{lb}}{\text{ft}^3}$

HORIZONTAL PLANE
 $z_1 = z_2 = z_3 = \text{same}$

- Find forces on nozzle
- Neglect friction ($\mu = 0$)

Energy ①-②: $\frac{P_1}{\rho} + \cancel{z_1} + \frac{V_1^2}{2g} + \cancel{h_{L,1-2}} = \frac{P_2}{\rho} + \cancel{z_2} + \frac{V_2^2}{2g}$

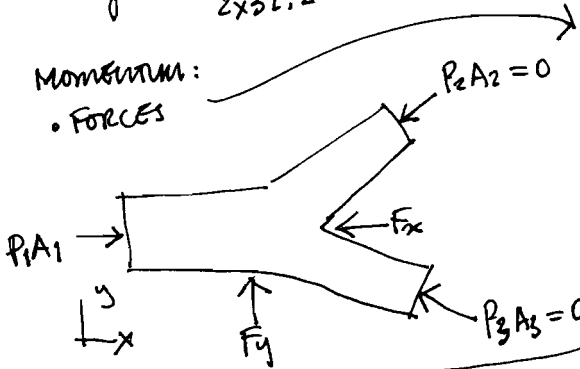
$\Rightarrow \frac{P_1}{\rho} = \frac{V_2^2 - V_1^2}{2g}$

continuity: $Q_1 = Q_2 + Q_3 \Rightarrow V_1 \cdot \frac{\pi D_1^2}{4} = V_2 \cdot \frac{\pi D_2^2}{4} + V_3 \cdot \frac{\pi D_3^2}{4}$

$V_1 = \frac{V_2 D_2^2 + V_3 D_3^2}{D_1^2} = \frac{40 \times 4^2 + 40 \times 6^2}{8^2} = 32.5 \text{ fps}$

$\frac{P_1}{\rho} = \frac{40^2 - 32.5^2}{2 \times 32.2} = 8.44 \text{ ft} \Rightarrow P_1 = 8.44 \times 62.4 = 526.86 \frac{\text{lb}}{\text{ft}^2}$

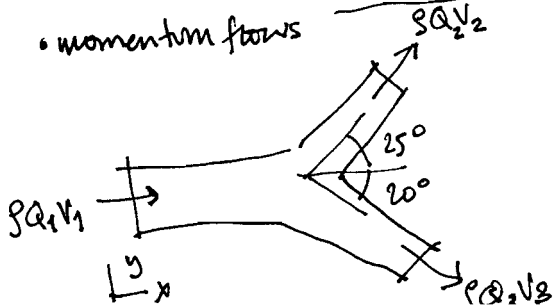
MOMENTUM:
 • FORCES



$\sum F_x = P_1 A_1 - F_x$
 $\sum F_y = F_y$

• momentum flows

$\Delta(\rho Q V)_x = \text{out-in}$
 $= \rho Q_2 V_{2x} + \rho Q_3 V_{3x} - \rho Q_1 V_{1x}$



$\Delta(\rho Q V)_y = \text{out-in}$
 $= \rho Q_2 V_{2y} + \rho Q_3 V_{3y} - \rho Q_1 V_{1y}$

discharges:

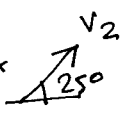
$$Q_1 = V_1 \frac{\pi D_1^2}{4} = 32.5 \times \frac{\pi (8/12)^2}{4} = 11.34 \text{ cfs}$$

$$Q_2 = V_2 \frac{\pi D_2^2}{4} = 40 \times \frac{\pi (4/12)^2}{4} = 3.49 \text{ cfs}$$

$$Q_3 = V_3 \frac{\pi D_3^2}{4} = 40 \times \frac{\pi (6/12)^2}{4} = 7.85 \text{ cfs}$$

$$\text{check, } \underline{Q_2 + Q_3 = Q_1}$$

velocity components:

$\rightarrow V_1$ $\left\{ \begin{array}{l} V_{1x} = V_1 = 32.5 \text{ fps} \\ V_{1y} = 0 \end{array} \right.$  $\left\{ \begin{array}{l} V_{2x} = V_2 \cos 25^\circ = 36.25 \text{ fps} \\ V_{2y} = V_2 \sin 25^\circ = 16.90 \text{ fps} \end{array} \right.$

$\searrow V_3$ $\left\{ \begin{array}{l} V_{3x} = V_3 \cos 20^\circ = 37.59 \text{ fps} \\ V_{3y} = -V_3 \sin 20^\circ = -13.68 \text{ fps} \end{array} \right.$

Components of momentum principle

$$\Sigma F_x = \Delta(\rho Q V)_x \Rightarrow P_1 A_1 - F_x = \rho (Q_2 V_{2x} + Q_3 V_{3x} - Q_1 V_{1x})$$

$$F_x = P_1 A_1 - \rho (Q_2 V_{2x} + Q_3 V_{3x} - Q_1 V_{1x})$$

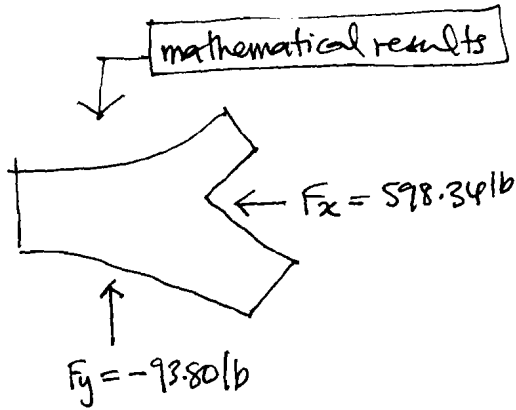
$$= 526.86 \times \frac{\pi (8/12)^2}{4} - \frac{62.4}{32.2} (3.49 \times 36.25 + 7.85 \times 37.59 - 11.34 \times 32.5)$$

$$\underline{F_x = 598.34 \text{ lb}}$$

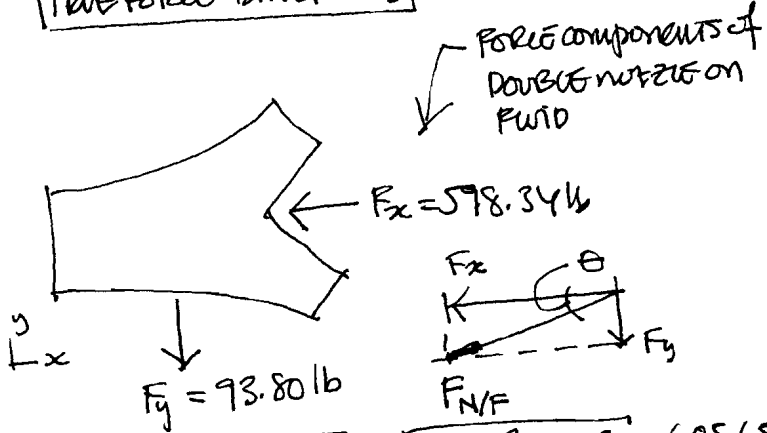
$$\Sigma F_y = \Delta(\rho Q V)_y \Rightarrow F_y = \rho (Q_2 V_{2y} + Q_3 V_{3y} - Q_1 V_{1y})$$

$$F_y = \frac{62.4}{32.2} (3.49 \times 16.90 + 7.85 \times (-13.68) - (11.34 \times 0))$$

$$\underline{F_y = -93.80 \text{ lb}}$$



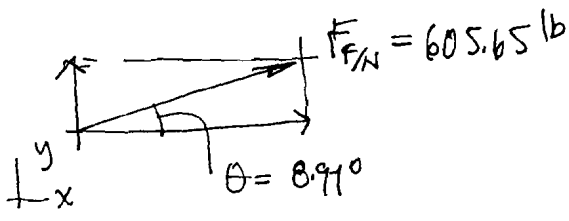
TRUE FORCE DIRECTIONS



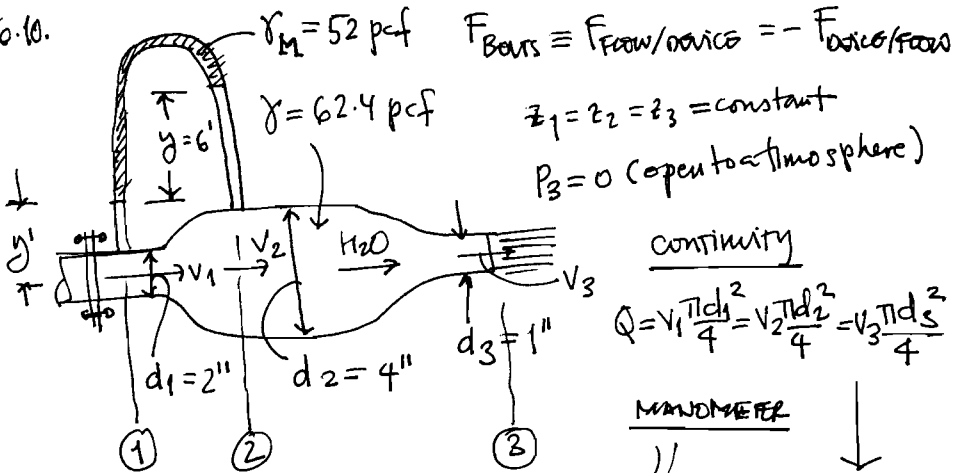
magnitude: $F_{N/F} = \sqrt{F_x^2 + F_y^2} = \sqrt{598.34^2 + 93.80^2} = 605.65 \text{ lb}$

$\theta = \tan^{-1}\left(\frac{F_y}{F_x}\right) = \tan^{-1}\left(\frac{93.80}{598.34}\right) = 8.91^\circ$

FORCE EXERCISED BY FLOW ON DOUBLE NOZZLE: same magnitude, but opposite direction as force from nozzle on flow, i.e., $\vec{F}_{R/N} = -\vec{F}_{N/F}$



6.10.



$$P_1 - \gamma y' - \gamma_m y + \gamma(y + y') = P_2 \Rightarrow \frac{V_1}{V_2} = \left(\frac{d_2}{d_1}\right)^2$$

$$P_2 = P_1 + y(\gamma - \gamma_m) = P_1 + 6 \text{ ft} (62.4 - 52) \frac{\text{lb}}{\text{ft}^3} = P_1 + 62.4 \frac{\text{lb}}{\text{ft}^2}$$

$$\text{Energy (1)-(2)}: \frac{P_1}{\gamma} + z_1 + \frac{V_1^2}{2g} = \frac{P_2}{\gamma} + z_2 + \frac{V_2^2}{2g} \Rightarrow \frac{P_2 - P_1}{\gamma} = \frac{V_1^2 - V_2^2}{2g}$$

(no losses)

$$\frac{P_2 - P_1}{\gamma} = \frac{V_1^2 - V_2^2}{2g} = \frac{V_2^2}{2g} \left[\left(\frac{V_1}{V_2}\right)^2 - 1 \right] = \frac{V_2^2}{2g} \left[\left(\frac{d_2}{d_1}\right)^4 - 1 \right]$$

$$V_2 = \sqrt{\frac{2g(P_2 - P_1)/\gamma}{(d_2/d_1)^4 - 1}} = \sqrt{\frac{2 \times 32.2 \text{ ft/s}^2 \times 1.0 \text{ ft}}{(4/2)^4 - 1}} = 2.07 \text{ ft/s}$$

$$\frac{P_2 - P_1}{\gamma} = \frac{62.4 \text{ lb/ft}^2}{62.4 \text{ lb/ft}^3} = 1.0 \text{ ft}$$

$$\text{Continuity: } V_1 = V_2 \left(\frac{d_2}{d_1}\right)^2 = 2.07 \text{ fps} \left(\frac{4}{2}\right)^2 = 8.28 \text{ fps}$$

$$V_3 = V_2 \left(\frac{d_2}{d_3}\right)^2 = 2.07 \text{ fps} \left(\frac{4}{1}\right)^2 = 33.12 \text{ fps}$$

$$Q = V_1 \frac{\pi d_1^2}{4} = 8.28 \frac{\text{ft}}{\text{s}} \times \frac{\pi \times (2/12 \text{ ft})^2}{4} = 0.1806 \text{ cfs}$$

$$\text{Energy (1)-(3)}: \frac{P_1}{\gamma} + z_1 + \frac{V_1^2}{2g} = \frac{P_3}{\gamma} + z_3 + \frac{V_3^2}{2g} \Rightarrow \frac{P_1}{\gamma} = \frac{V_3^2 - V_1^2}{2g}$$

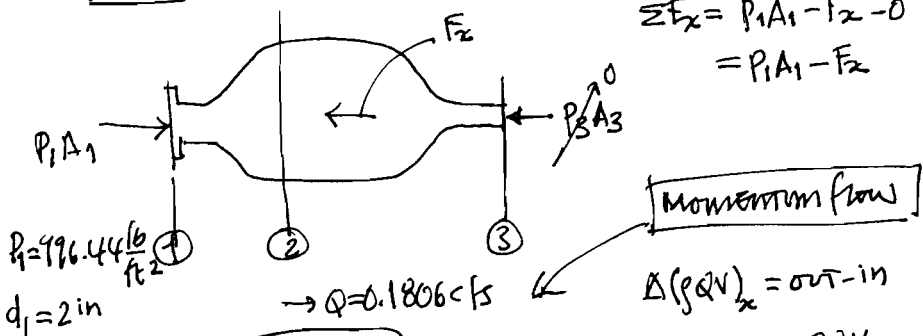
(no losses)

$$P_1 = \frac{\gamma}{2g} (V_3^2 - V_1^2) = \frac{62.4 \text{ lb/ft}^3}{2 \times 32.2 \text{ ft/s}^2} \times (33.12^2 - 8.28^2) \frac{\text{ft}^2}{\text{s}^2} = 996.44 \text{ lb/ft}^2$$

$$\text{also, } P_2 = P_1 + 62.4 \frac{\text{lb}}{\text{ft}^2} = (996.44 + 62.4) \frac{\text{lb}}{\text{ft}^2} = 1058.84 \frac{\text{lb}}{\text{ft}^2}$$

MOMENTUM EQUATION

Forces F_x : force from device on flow



momentum equation: $\Sigma F_x = \Delta(\rho Q V)_x$

$$P_1 A_1 - F_x = \frac{\gamma}{g} Q (V_3 - V_1) \Rightarrow F_x = P_1 A_1 - \frac{\gamma}{g} Q (V_3 - V_1)$$

$$F_x = 996.44 \frac{\text{lb}}{\text{ft}^2} \times \frac{\pi \times (2/12 \text{ ft})^2}{4} - \frac{62.4 \text{ lb/ft}^3}{32.2 \text{ ft/s}^2} \times 0.1806 \frac{\text{ft}^3}{\text{s}} \times (33.12 - 8.28) \frac{\text{ft}}{\text{s}}$$

$$F_x = 13.05 \text{ lb} \leftarrow \text{FORCE FROM DEVICE ON FLOW}$$

Force of flow on device: $F = 13.05 \text{ lb} \rightarrow$

